



# ***Setting the Standard: Recommendations on “Launch Unit” Standard SmallSat Sizes between CubeSats and ESPA-Class***

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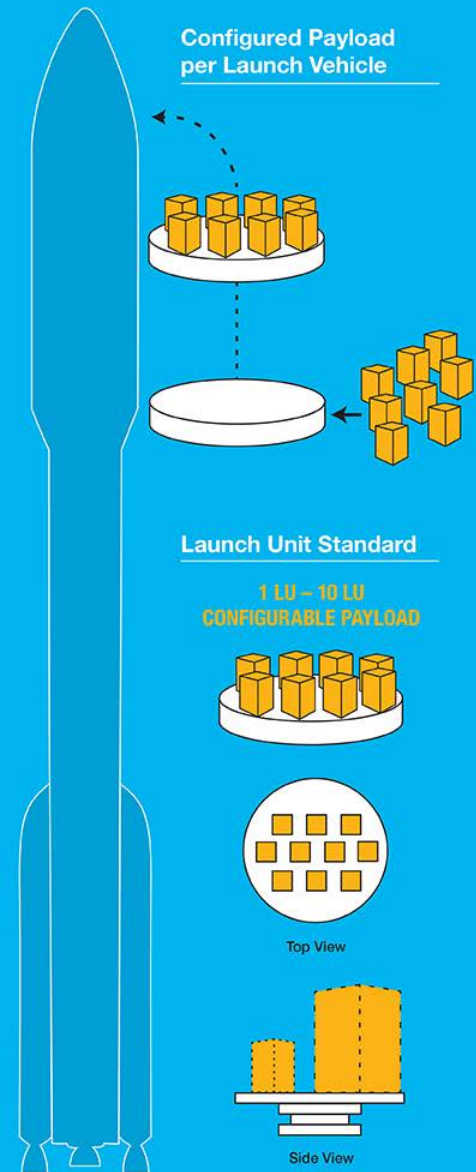


# Small Satellites are Abundant

*How do they all get to space?*

Fewer than 700 small satellites were launched between 2006 and 2015

Over the next 10 years, estimates are an additional 3,500 to 10,000 satellites will be launched<sup>1</sup>



0.25 kg



3.5 kg



164 kg

Small satellites have a mass less than 500 kg

- *CubeSats* are a special subset of small satellites which conform to the “CubeSat Standard”
- The “ESPA class” is an informal standard for a satellite with a mass of approximately 180 kg

## Small satellites are on the rise!

1. Euroconsult, “Prospects for the Small Satellite Market,” (Jul. 2017)



# Traditional Rideshare vs Rideshare with LaunchU



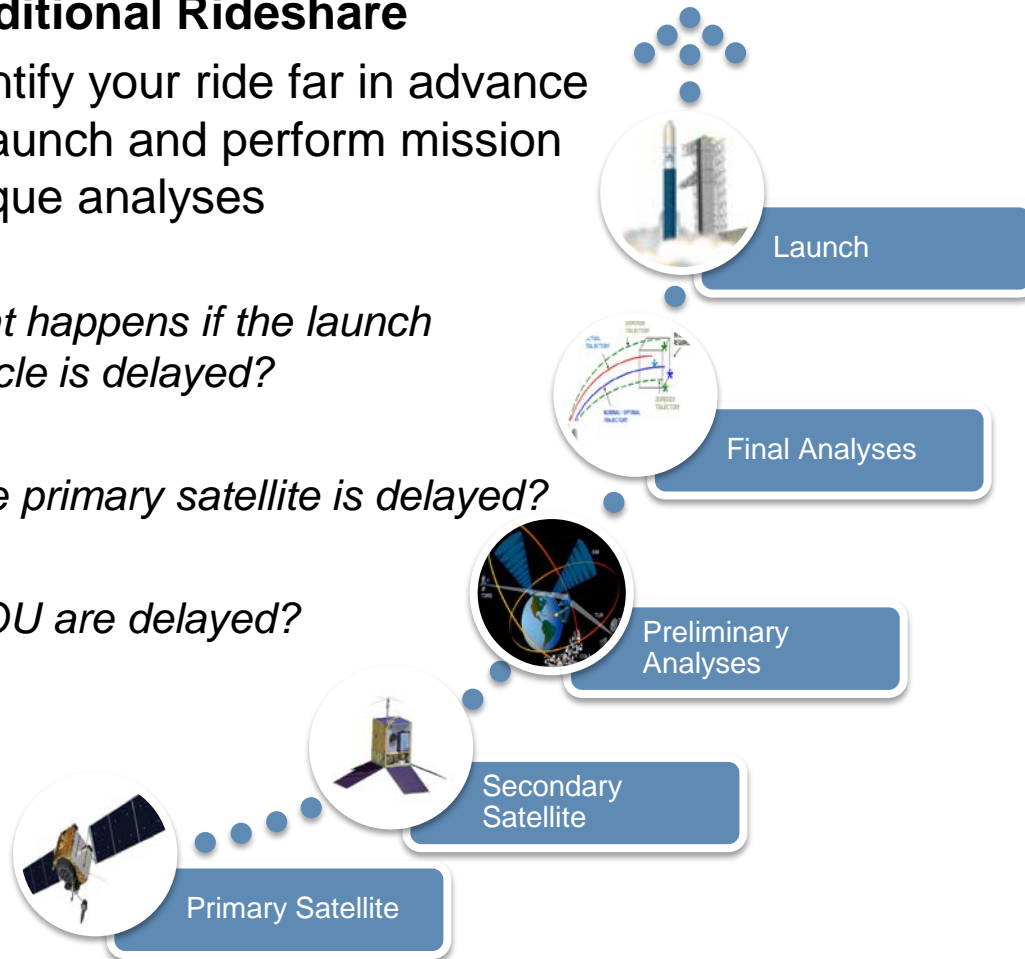
## Traditional Rideshare

Identify your ride far in advance of launch and perform mission unique analyses

*What happens if the launch vehicle is delayed?*

*If the primary satellite is delayed?*

*If YOU are delayed?*

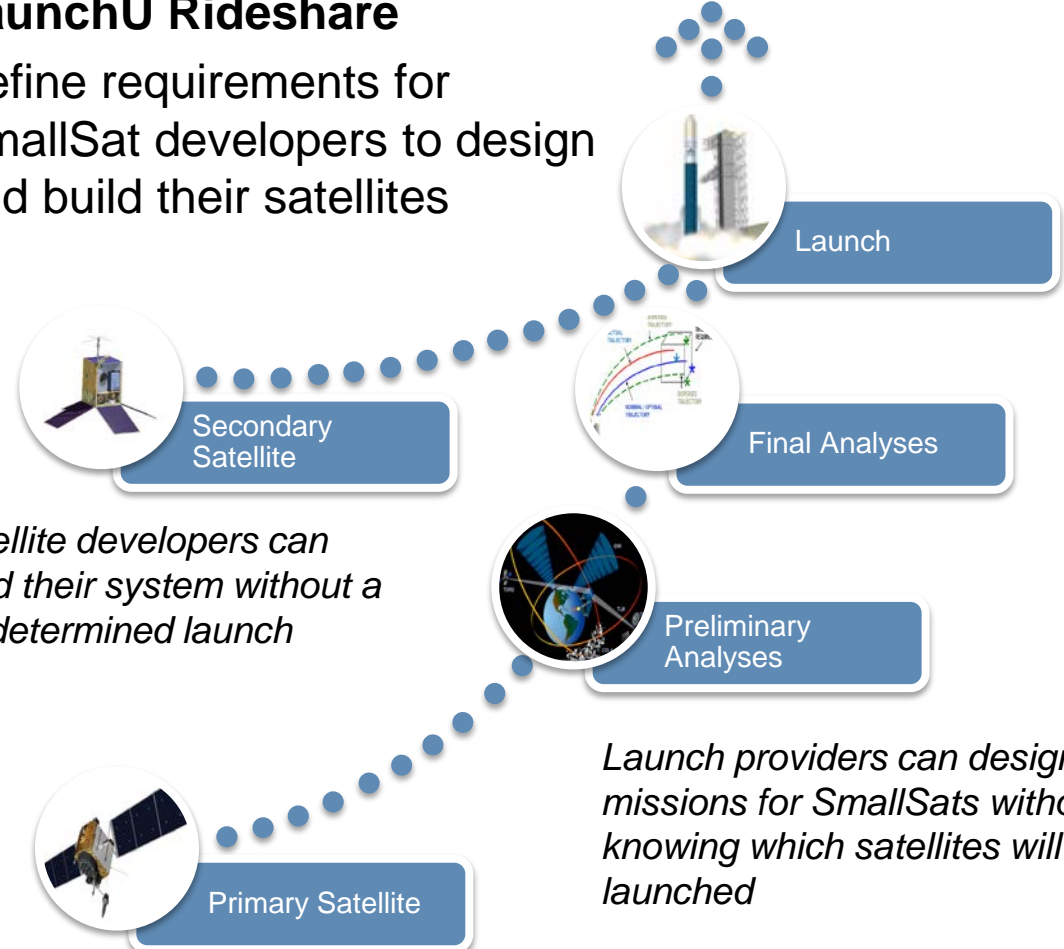


## LaunchU Rideshare

Define requirements for SmallSat developers to design and build their satellites

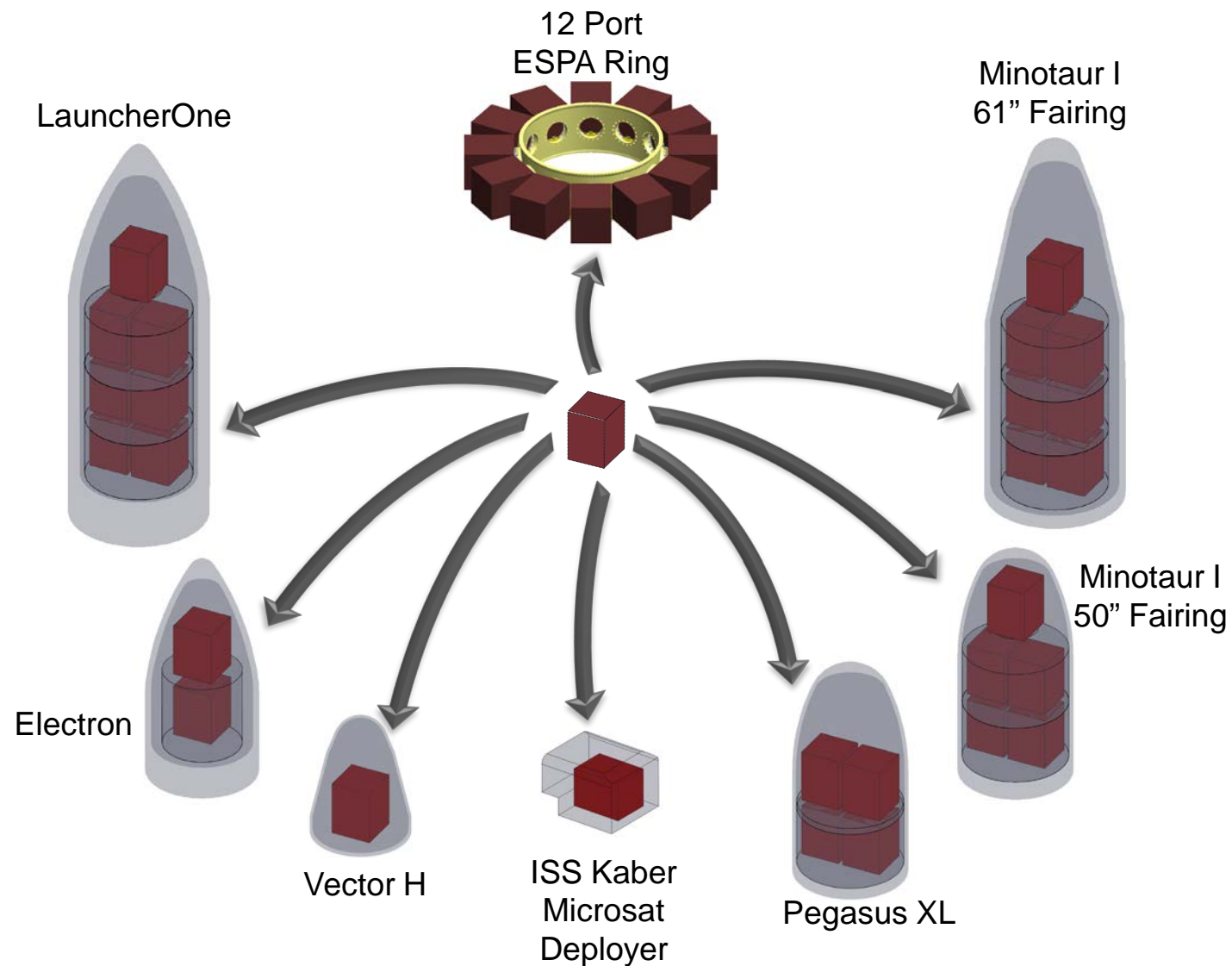
*Satellite developers can build their system without a predetermined launch*

*Launch providers can design missions for SmallSats without knowing which satellites will be launched*

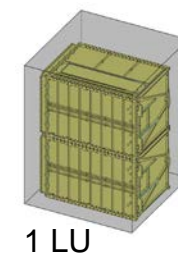


**LaunchU enables flexible assured access to space for the small satellite market**

# Volume Efficiency

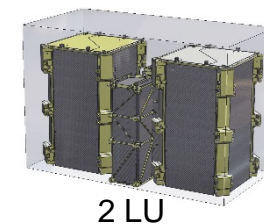


## Alternate Uses of LaunchU Volume



2 Planetary Systems Corporation (PSC) 12U Dispensers

2 12U and 2 3U Tyvak Dispensers



***LaunchU provides a more efficient use of launch vehicle fairing volume for small satellites***

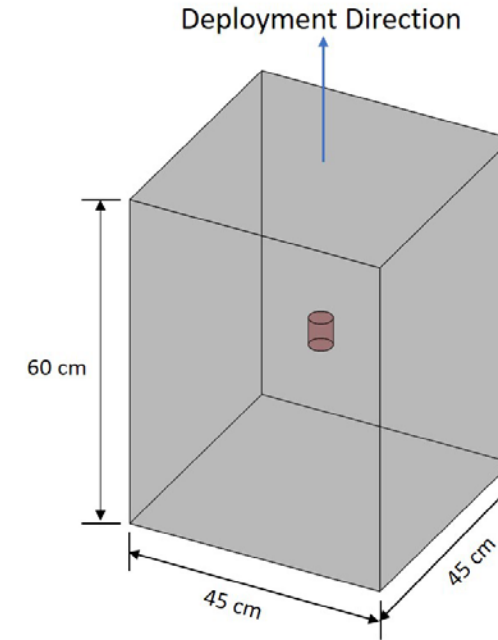
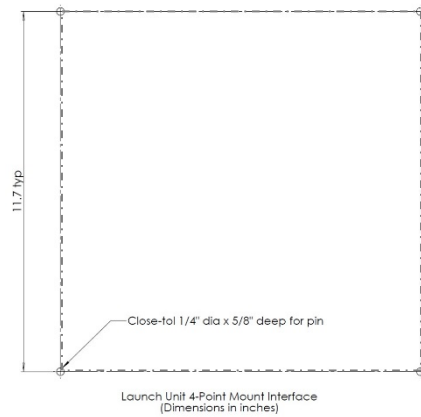
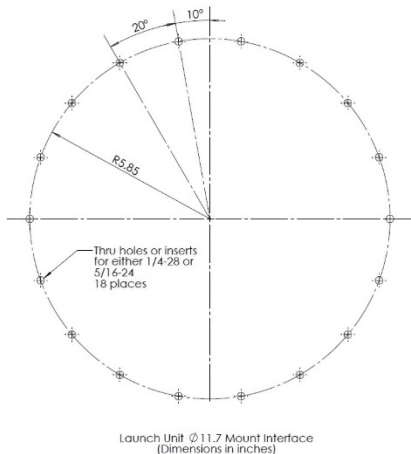
# LaunchU Recommendations

## Volume, Mass, and Dynamics

### INCLUDES SEPARATION SYSTEM

- Volume: 45 x 45 x 60 cm
- Mass: 60-80 kg
- CG:  $30 \pm 5$  cm along height, within 2 cm of centerline
- First fundamental frequency: above 50 Hz in both axial and lateral directions

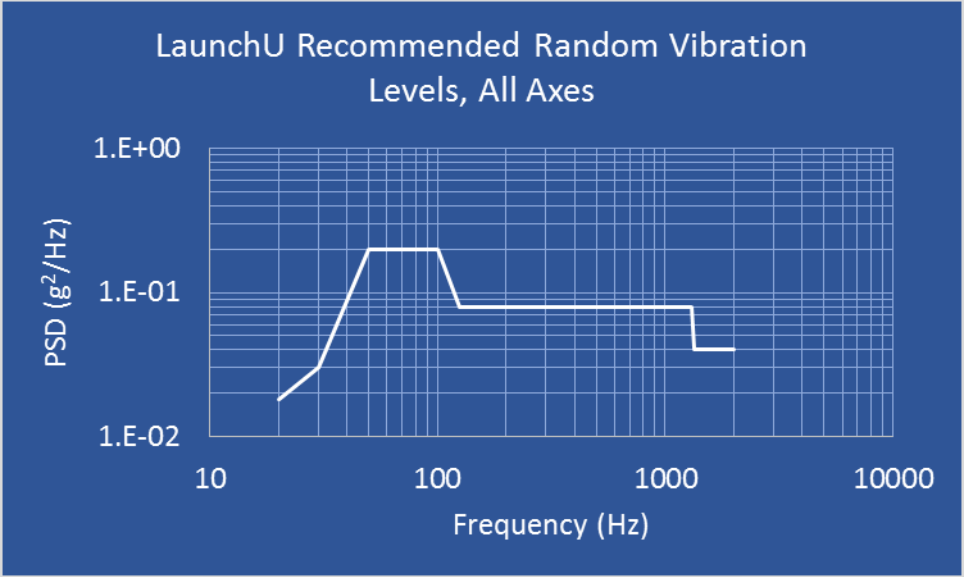
### • Mechanical Interfaces



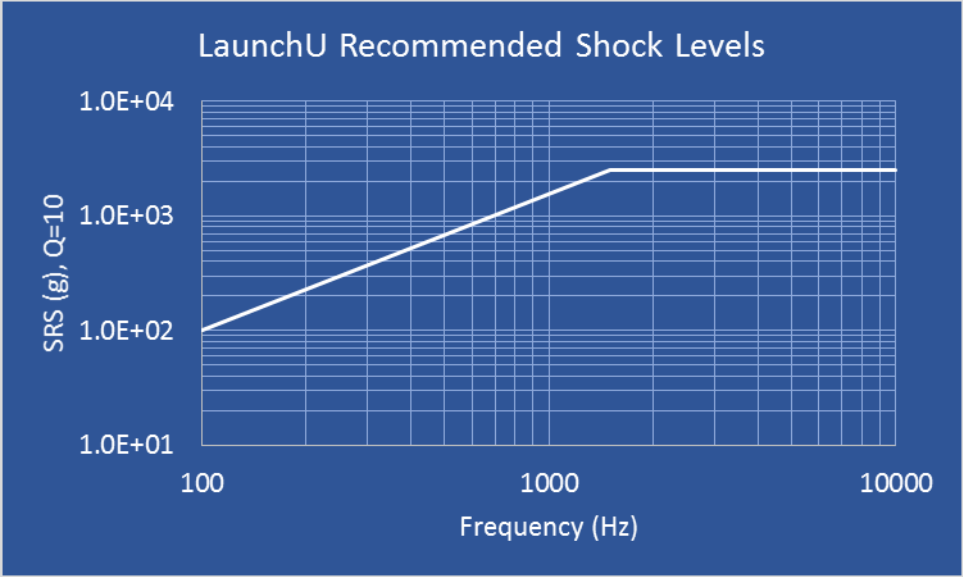
### Electrical Interfaces

- Limited to
  - Signals required to initiate separation system (sent by launch vehicle)
  - Loopback circuits for separation indication
- No LaunchU satellite telemetry data transmitted via the launch vehicle telemetry units
- LaunchU satellite telemetry transmitters will not radiate prior to separation

# Load Recommendations



| Frequency (Hz) | Acceptance Level PSD (g²/Hz) |
|----------------|------------------------------|
| 20             | 0.018                        |
| 30             | 0.03                         |
| 50             | 0.2                          |
| 100            | 0.2                          |
| 125            | 0.08                         |
| 1300           | 0.08                         |
| 1350           | 0.04                         |
| 2000           | 0.04                         |
| gRMS           | 11.76                        |



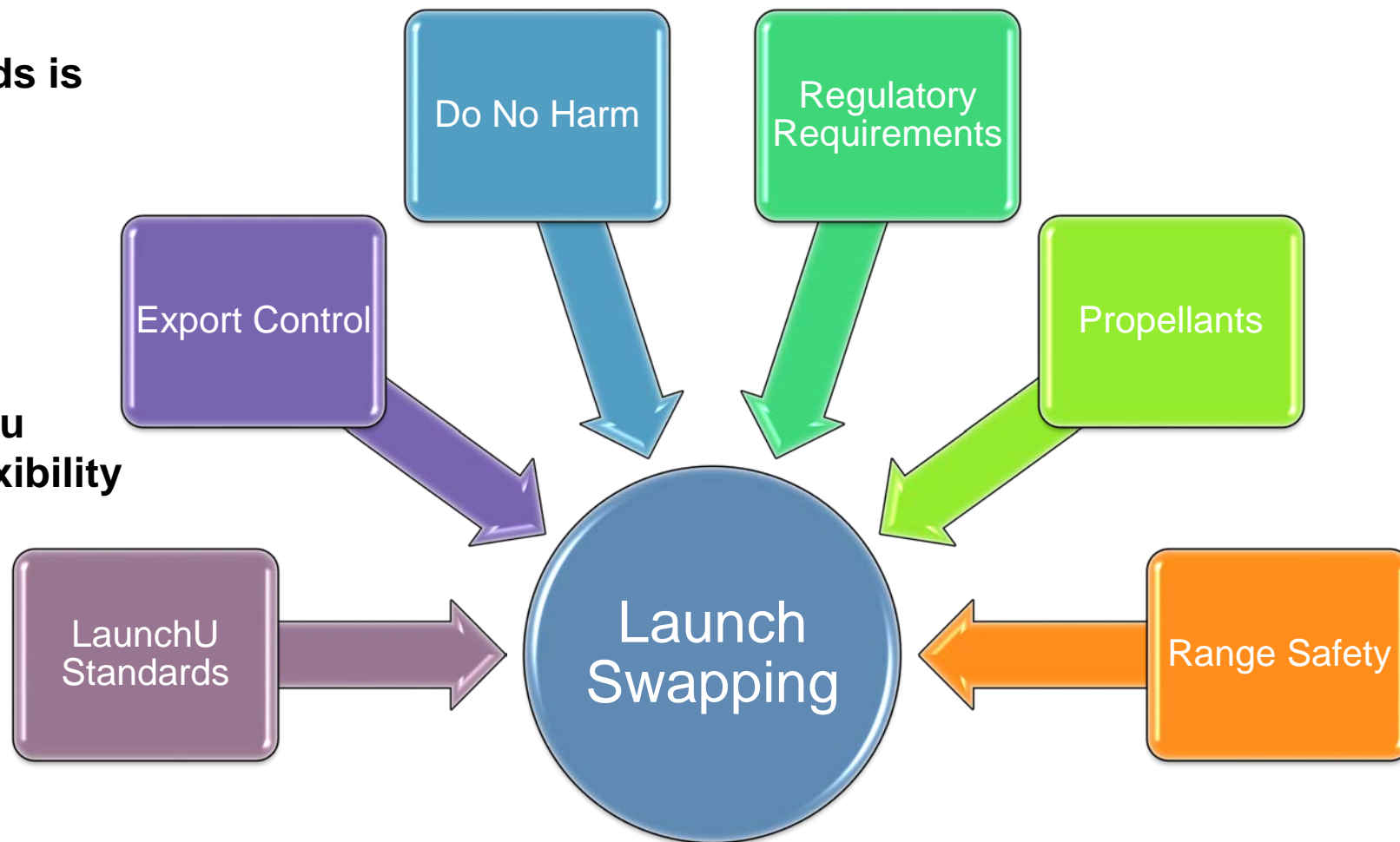
| Frequency (Hz) | Shock Response Spectrum (g) |
|----------------|-----------------------------|
| 100            | 100                         |
| 1500           | 2500                        |
| 10000          | 2500                        |
|                | Q=10                        |

Static Loads: 8.5 Gs applied at the center of gravity in each direction

# Additional Considerations

**Meeting the LaunchU standards is only part of the story**

**By addressing other areas, you dramatically increase your flexibility in launch opportunities**



## Next Steps

Engage SmallSat community on recommendations

## Feedback Needed!

- Details provided in SSC-IX-07
- Come see me and ask questions today in the Aerospace booth (#166) from 4-5 pm
- Visit [www.aerospace.org/launch-u](http://www.aerospace.org/launch-u) and provide feedback

Pursue formal LaunchU standard





# Launch Unit Consortium Members



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